

## ADAPTIVE AEROMAGNETIC REAL-TIME COMPENSATION FOR UAVs

*Magnetometer interface  
counter for one sensor*

*Real-time compensation*

*Magnetics & general-purpose  
data acquisition & recording*

*Built-in interface for external  
GPS receiver*

*Compact, light, low power –*

**133 x 133 x 216 mm**

**+28 VDC (19–36 VDC), 3 A**

**< 1.8 Kg**

- Front-end sampling rates up to 1280 Hz
- Magnetometer processor: 0.32 pT resolution, < 0.1 pT system noise,  $\pm 10$  ppb temp. stability
- Proven, highly-robust compensation algorithms (AADCI legacy)
- Adaptive signal processing techniques
- User may customize Front End processing to specific installation requirements
- Interface for external GPS receiver
- Recording up to 80 Hz: complete raw, pre-processed and compensated data sets
- Post-flight compensation & analysis functions
- Dynamic compensation of on-board electronics
- Real-time graphical output and user interface on any external display
- Two general-purpose analog inputs (e.g., radar altimeter), two assigned to embedded barometric pressure & temperature sensors
- Data acquisition via Ethernet
- Full monitoring/control from any Windows device (via Ethernet, or remotely via internet)
- Features targeted to UAV applications
- State-of-the-art HW & FW architecture based on advanced 64/32-bit processors
- Real-time operating system (RTOS): QNX 6.5



## Aeromagnetic Compensation

The quality of the data collected in aeromagnetic surveys is largely dependent on the quality of compensation. Despite the outstanding sensitivity of modern magnetometers, in the absence of good compensation anomaly signals can be completely masked out by the interference of the nearby magnetics of the aircraft.

The aircraft's magnetic interference is related to its motions about its principal axes. A mathematical model may be built to accurately represent the aircraft's magnetic signature. Careful optimization and implementation of this model, within the framework of sophisticated hardware and firmware technologies, can lead to real-time compensation that effectively eliminates the aircraft's magnetic interference.

The RMS Instruments' AARC51 Adaptive Aeromagnetic Real-Time Compensator provides real-time compensation of local magnetic interference for inboard magnetometer systems in fixed-wing aircraft, helicopters, or UAVs to the point where the full resolution of modern high sensitivity magnetometers can be utilized. The compensation accounts for the effects of permanent and induced magnetism, eddy currents, and heading errors from the sensors.

### The importance of real-time compensation

State-of-the-art aeromagnetic surveying requires real-time monitoring of compensated data, so that problems are identified immediately and are promptly corrected. The magnetic signature of typical survey aircraft can be rather unstable and change in-flight; detecting these changes while monitoring uncompensated data is practically impossible. By eliminating costly and error-prone post-flight processing, real-time compensation further contributes to increased productivity, efficiency and cost-effectiveness.

### Calibration mode, model and solution

The AARC51 uses a 3-axis fluxgate magnetometer to monitor the aircraft's position and motion with respect to the ambient magnetic field while flying a set of standard maneuvers of rolls, pitches and yaws in orthogonal headings. This *calibration* process, which typically takes 6-8 minutes, yields a (mathematical) *solution* that models the aircraft's magnetic signature. The solution is calculated instantly, upon termination of the calibration maneuvers. It is immediately available for use in compensated (i.e., survey) mode, or for further analysis and comparison with other solutions.

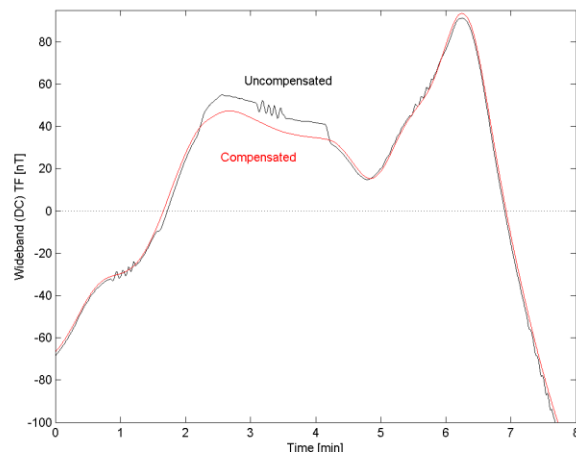
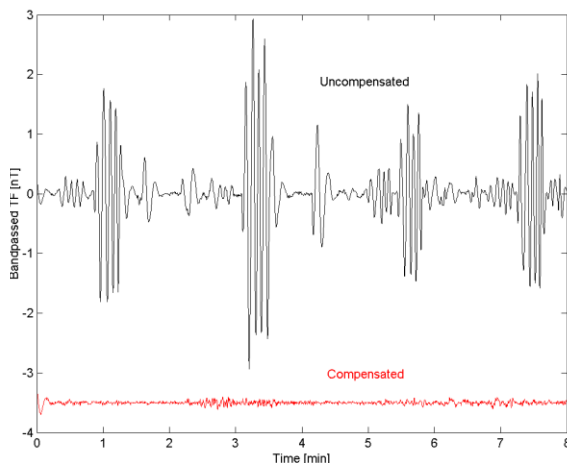
With the AARC51 there is no need for any post-flight software. The calibration is effective for the full 360° range of headings. At very low dip angles, *partial calibrations* for each active zone can be readily combined to produce a single robust solution for the full 360° range.

### Compensation mode

In *compensation* mode the total-field (TF) high-sensitivity magnetometer is compensated in real-time using the last solution obtained (or any other solution previously archived). Compensated and uncompensated data, along with the 3-axis vector magnetometer and other ancillary data, are monitored and recorded in real-time.

### Adaptive compensation

The AARC51 incorporates adaptive signal processing techniques that allow the system to continuously "learn" from input signals, and adapt the solution coefficients for optimum compensation. The underlying recursive algorithm has significant computational advantages over the "conventional" alternative, and leads to improved band-passed compensation. Adaptive compensation substantially eases calibration procedures, and yields solutions that remain close to optimum as the aircraft's magnetic signature changes with time.



Left – Bandpassed uncompensated and compensated data for a full calibration flight (8 minutes). The uncompensated waveform clearly shows the aircraft interference on the four headings.

Performance indicators:  $\sigma_{uncomp} = 0.5502 \text{ nT}$ ,  $\sigma_{comp} = 0.0282 \text{ nT}$ ,  $IR = 19.5$ . (Waveforms are offset for clarity.)

Right – Wideband uncompensated and compensated waveforms. (Mean value subtracted for clarity.)

## Dynamic compensation of OBE systems

The AARC51 incorporates new technology that allows real-time dynamic compensation of the effects of DC currents from on-board electronic (OBE) systems, such as avionics, hydraulics, control systems and other instrumentation. OBE compensation simplifies operational requirements, increases robustness and tolerance to electrical sources, and improves overall compensation performance. The technology works with fixed- and variable-current devices, for up to two independent OBE systems.

## System Architecture

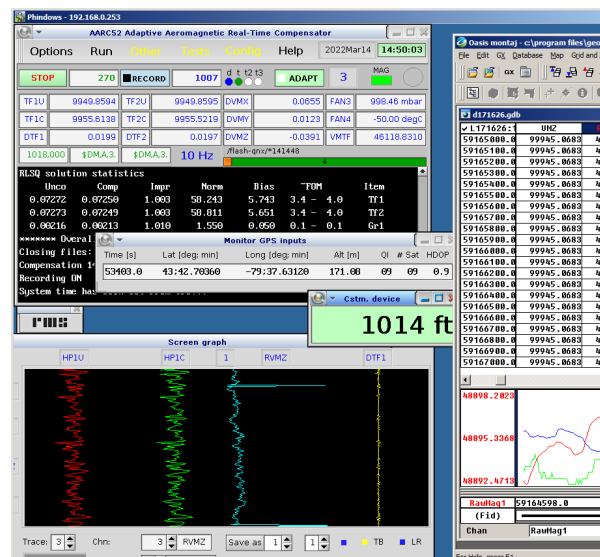
RMS Instruments' compensation technology is based on a flexible architecture with dual 64/32-bit processors. It includes state-of-the-art COTS (industrial-grade) electronics, and a proprietary magnetometer interface with excellent accuracy.

Front End sampling rates up to 1280 Hz and finely-tuned transfer functions deliver outstanding anti-aliasing characteristics, and may be customized by the user to the specific requirements of an installation.

The main program and real-time operating system (RTOS) reside in solid-state memory. The RTOS is QNX 6.5 (or later), a deterministic and extremely reliable operating system tailored to mission-critical applications.

A three-axis fluxgate (vector) magnetometer is included with the system. Signals are processed using a high-resolution (16-bit) A/D converter.

The software includes an easy-to-use graphical user interface, and a rich set of utilities to analyze data and help troubleshoot aeromagnetic installations.



## Remote control from Windows

A remote connectivity tool for the AARC51 allows users full control and operation of the unit from a remote Windows-based system, across an IP network.

The user interface of the AARC51 is seamlessly replicated in the Windows-based computer. The mouse and keyboard attached to the computer have the same effect as if they were directly connected to the system.

This technology facilitates integration of complex systems, with a single computer/laptop being used to con-

trol and operate the AARC51 and other instruments, while simultaneously running complementary software.

## GPS receiver

An interface for an external GPS receiver is standard. Any receiver with NMEA GGA output via a serial port (RS232, 115.2 kbps, up to 10 Hz) and PPS trigger pulse may be used. GPS data are appended to recorded and transmitted magnetics data packets. Timing throughout the system is tied to the PPS trigger from the GPS receiver.

## Post-flight compensation & analysis

Advanced embedded functions allow post-flight survey compensation, in the event a suitable calibration was not available at time of flight. This complements the fundamental real-time compensation function, key for productive and efficient airborne magnetometry. Also included are functions for in-depth analysis of calibration data, and frequency-domain analysis.

## General-purpose data acquisition

The AARC51 supports data acquisition via Gigabit-Ethernet (with multiple logical connections), and includes two general-purpose differential, high-resolution analog inputs (e.g., radar/laser altimeter) and two dedicated to embedded pressure and temperature sensors.

The system includes flexible means for real-time monitoring of the data streams acquired via Ethernet, as well as the analog inputs.

## UAV Applications

The AARC51 is ideally suited to Unmanned Aerial Vehicle (UAV) applications because of its light weight, compact package, and low power consumption.

In typical UAV use the unit is pre-configured on the ground at the start of a project, set up to automatically start data acquisition, compensation and recording on power up. This is easily done using a laptop/computer via Ethernet, or by attaching a display and mouse/keyboard. During surveys the AARC51 is essentially a "black box" continuously recording in an embedded solid-state drive (Flash) magnetics data (raw and compensated), GPS and other ancillary data. Upon landing, the data are extracted from the system through a USB memory stick or via Ethernet.

For optimum performance a suitable calibration flight must be carried out as far as possible from any geological and/or cultural interference. Without an operator on-board this presents unique challenges in UAV systems. The AARC51 allows real-time command and control from the ground with a long-range wireless Ethernet link onboard the UAV. Alternatively, it provides two comprehensive approaches to automate the calibration process: time-framed calibrations (TFC), and altitude-controlled calibrations (ACC). With TFCs the operator pre-defines a specific time window after power up, during which calibration maneuvers must be flown. With ACCs the start/end of calibrations may be controlled through the embedded barometric pressure sensor (or an external sensor/altimeter), or GPS altitude.

## AARC51 SPECIFICATIONS

### Magnetometer Inputs:

One high-sensitivity magnetometer:

Cs: Typ. 70 kHz – 350 kHz

### Magnetic Field Range:

Per the magnetometer's range; e.g.: <sup>[1]</sup>

G-822A, G-823A: 20,000–100,000 nT

CS-3, CS-L, CS-VL: 15,000–105,000 nT

### Front End (FE):

Time base: > 100 MHz, OCXO

Resolution: 0.32 pT <sup>[2]</sup>

System noise:  $\sigma < 0.1$  pT <sup>[3]</sup>

Temperature stability:  $\approx \pm 10$  ppb <sup>[4]</sup>

Sampling rate: 160, 640, 800 or 1280 Hz  
– user-selectable

Transfer function (bandwidth): 1.6 Hz,  
3.25 Hz, 6.4 Hz, 9.8 Hz, 20 Hz,  $0.16F_{SH}$   
or Custom – user-selectable

### Compensation Performance:

Conventional (manned) fixed-wing, helicopter:

$\sigma \approx 20$  pT, full flight envelope, 0–1 Hz

IR, total field (TF): 10–20, typical

Adaptive mode: Recursive approach  
with user-selectable gain; up to 2–5X  
further improvement (typical), band-  
passed TFs.

UAV: Consult RMS Instruments

### Optional Filter (Host):

User-selectable, 0.4 – 3.0 Hz BW

### Calibration Duration:

6–8 minutes, typical

### Vector Magnetometer:

Included with the AARC51

3-axis fluxgate

Oversampling, 16-bit, self-calibrating ADC

### OBE Compensation:

Dynamic compensation of up to 2 independent on-board electronic systems

### Data Output & Recording:

$F_{SH}$ : 10, 20, 40, 80 Hz (GPS-PPS or internal synch.); external-trigger

Ethernet: TCP/IP packets, ASCII/Binary

Recording media: embedded Flash SSD  
( $\geq 8$  GB), USB Flash disk

External display (VGA)

### Event Inputs:

PPS trigger signal from external GPS

Two general-purpose latched event inputs

LS-TTL levels, edge-sensitive

Event tags included with output data

Accuracy: per Front End sampling rate

### Embedded Barometric Pressure & Temperature Sensors:

Differential inputs, 16-bit ADC

600 to 1100-mbar range;  $\pm 5$  mbar total accuracy

–50 to +100°C range;  $\pm 1^\circ\text{C}$  abs. error

### Raw Data Logging:

At Front End sampling rate

Single 1-MB buffer burst

### FE-Sampled Analog:

Two differential inputs

16-bit resolution, self-calibrating ADC

Input range:  $\pm 5$  Volts

Input resistance: 1 M $\Omega$ , typical

### Data Acquisition via Ethernet:

10/100/1000Base-TX

Real-time, streaming data (TCP/IP)

Two independent logical connections

Sampling & recording:  $F_{SH}$  or submult.

### Remote Control:

Optional: From any Windows-based computer, via Ethernet – full replica of user I/F

### Indicators, General-Purpose I/O:

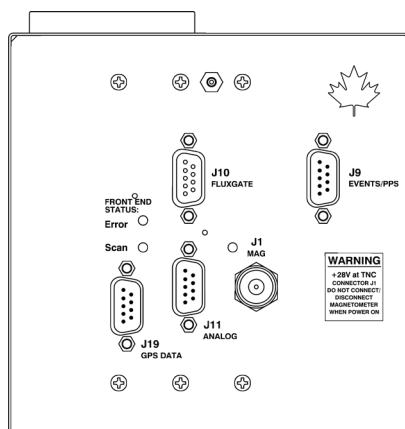
1 LEDs for mag. input status

2 LEDs for Front End status

Three USB 2.0

10/100/1000Base-TX Ethernet (RJ45)

VGA video (15-pin D-sub)



### GPS Receiver:

Standard: Interface to any GPS receiver with NMEA GGA output via serial port (RS232, 115.2 kbps, up to 10 Hz), and PPS trigger (LS-TTL or 5V-tolerant LV-TTL)

Magnetics data tagged with GPS time, lat., long., altitude, and auxiliary data  
Up to 10 Hz

### Post-Flight Compensation:

Advanced analysis functions on standard system-recorded d-files:

Post-flight compensation

Calibration/solution robustness analysis

Frequency-domain analysis

### Power:

Nominal: +28 VDC, 3.0 A

(Total power requirement, including magnetometer sensor <sup>[5]</sup>)

Range: +19 to +36 VDC

Absolute maximum: +50VDC, < 100 msec

### Environmental:

Operating Temperature: –10 to +50°C  
(0 to +50°C with HW Rev. < 3.00)

Storage Temperature: –20 to +55°C

Relative Humidity: 0 to 99%, non-condensing

Altitude: 0–6,000 m (0–20,000 ft)

Size (W x H x D): 133 x 133 x 216 mm,  
(5¼ x 5¼ x 8½ in)

Weight: < 1.8 Kg

(Excluding external cables,  
mounting hardware & sensors)

### Notes:

[1] Per manufacturer's specs. at print time: G-822A, G-823A (Geometrics), CS-3, CS-L, CS-VL (Scintrex).

[2] 1.6-Hz BW, 625-ms integ.

[3] Typical; 1.6-Hz BW, 625-ms integ., 10 Hz.

[4] Within operating temperature range. Over –20 to +70°C,  $\leq \pm 35$  ppb.

[5] Per Geometrics G822A-type sensors. May vary for other sensors.

Specifications per Host FW  $\geq$  RMS11093-02-E,  
FE FW  $\geq$  RMS1877-05-B, HW Rev.  $\geq$  3.10,  
and subject to change without notice – Jun 2023.

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